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A New Mechanistic Design Procedure for Flexible Airfield Pavements

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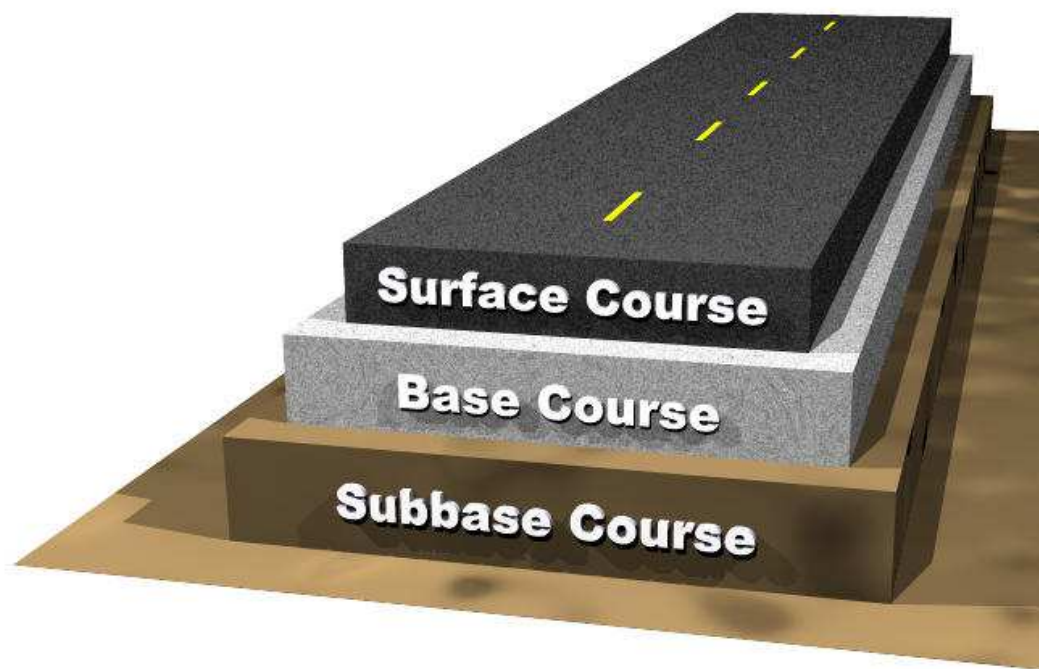
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What is pavement design?

- Determination of the thickness of the layers constituting the pavement structure for a given traffic mix

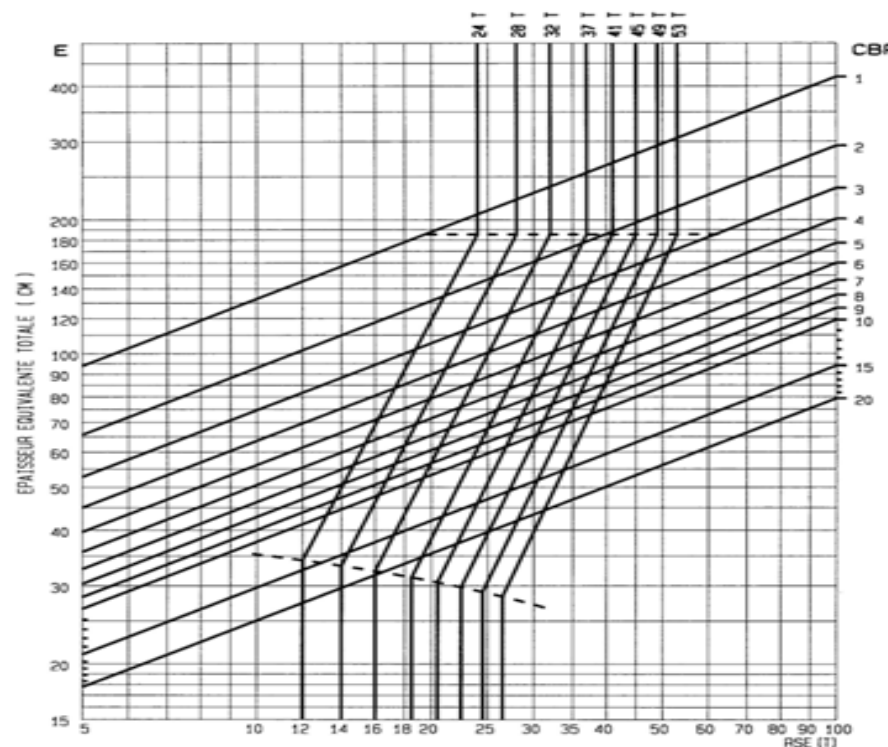




A necessary revision



- Former empirical method (DCA software)
 - Inspired by the US Corps of Engineers design method
 - Based on the CBR method





A necessary revision



- Many limitations
 - No explicit consideration of asphalt material damage
 - CBR tests for soil's characterization
 - New materials not correctly considered
 - No modeling of binding conditions between layers
 - No consideration of temperature and speed
 - Not appropriate for new landing gear configurations





New methodology



- Based on a more rational approach
- Used in France for roads for more than 30 years
- Development of the airfield rational design method from the road rational design method
- Restricted to new flexible airfield pavement first
- Working group: STAC – ISFTTAR – CEREMA



IFSTTAR





Design manual



- Published in January 2014
- www.stac.aviation-civile.gouv.fr
- Available in English soon





Design guide

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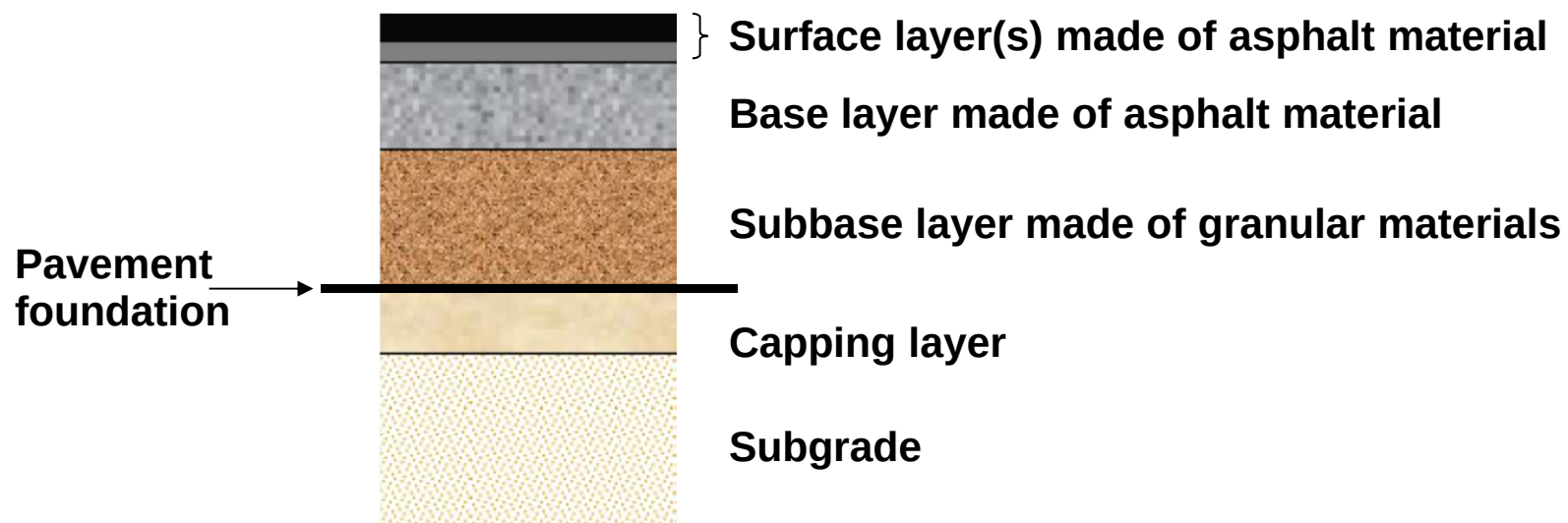


- Introduction
- Design method principle
- Implementation of the method
- Capping layer and subgrade
- Surface layer
- Pavement materials
- Freeze / thaw verification
- Application examples



Scope of the guide

- New flexible pavements



- Used for the design of all the sections of an airport: runways, taxiways and aprons



Calculation steps

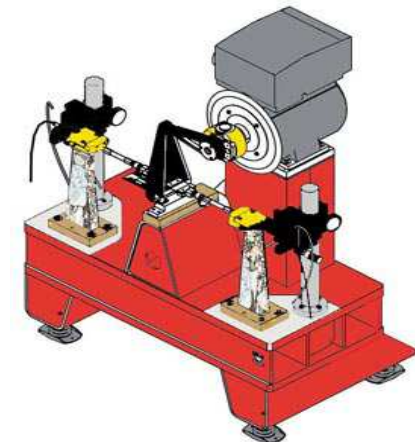


- 1- Gathering of all the input data
- 2- Calculation of the strains and stresses
- 3- Determination of the damage (D=0 new pavement, D=1 pavement failed)
 - Without and with lateral wander
 - Cumulated over the design period
- 4- Iterations on the pavement layers thickness until D equals 1 (must comply with technological limits)
- 5- Freeze-thaw verification



Input data

- Design life: 10 years (20 years for rigid pavements)
- Design temperature (equivalent temperature):
 - Oceanic, Mediterranean or continental climate: 15°C
 - Tropical climate: 28°C
- Risk coefficient r : to be set by the airport manager
 - 2.5% for high traffic, between 5 and 10% for low traffic
- Materials to be used and their fundamental properties:
Complex modulus, fatigue resistance...





Input data



- Subgrade bearing capacity
 - Assessed using static or dynamic plate test



- Elastic modulus (direct input of the design method)

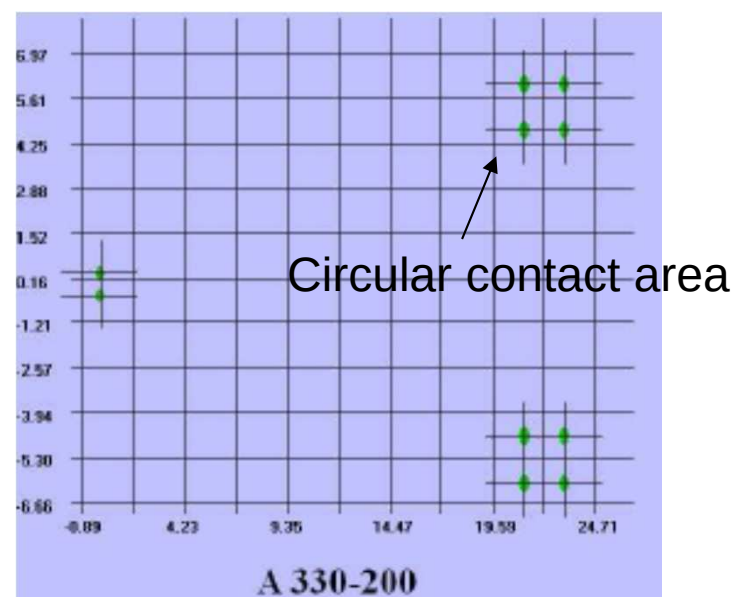


Input data

■ Traffic inputs:

- Type of aircrafts
- Number of passes (with distinction between take-offs and landings)
- Landing gear configuration and loadings: coordinates of the wheels center, mass on each wheel, contact pressure (supposed to be equal to the tyre pressure)
- Maximum ramp weight of the aircraft for take-offs
- Maximum landing weight of the aircraft for landings

STAC's "Ficav" database





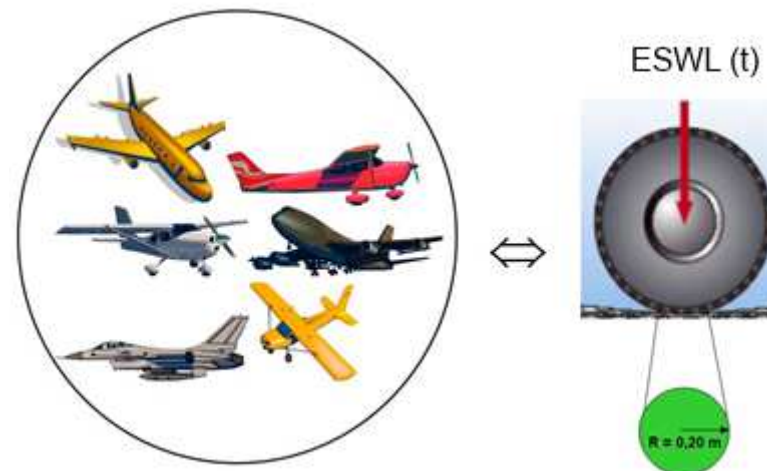
Input data



- Traffic aggressiveness quantified by the rational Equivalent Single Wheel Load (ESWL)

«The ESWL associated with an aircraft traffic mix and a pavement structure is the simple, non-wandered load (in tons) applied 10,000 times to the structure, with a footprint with a radius of 0.20 m, which produces the same value of fatigue damage of the asphalt concrete as the complete traffic mix »

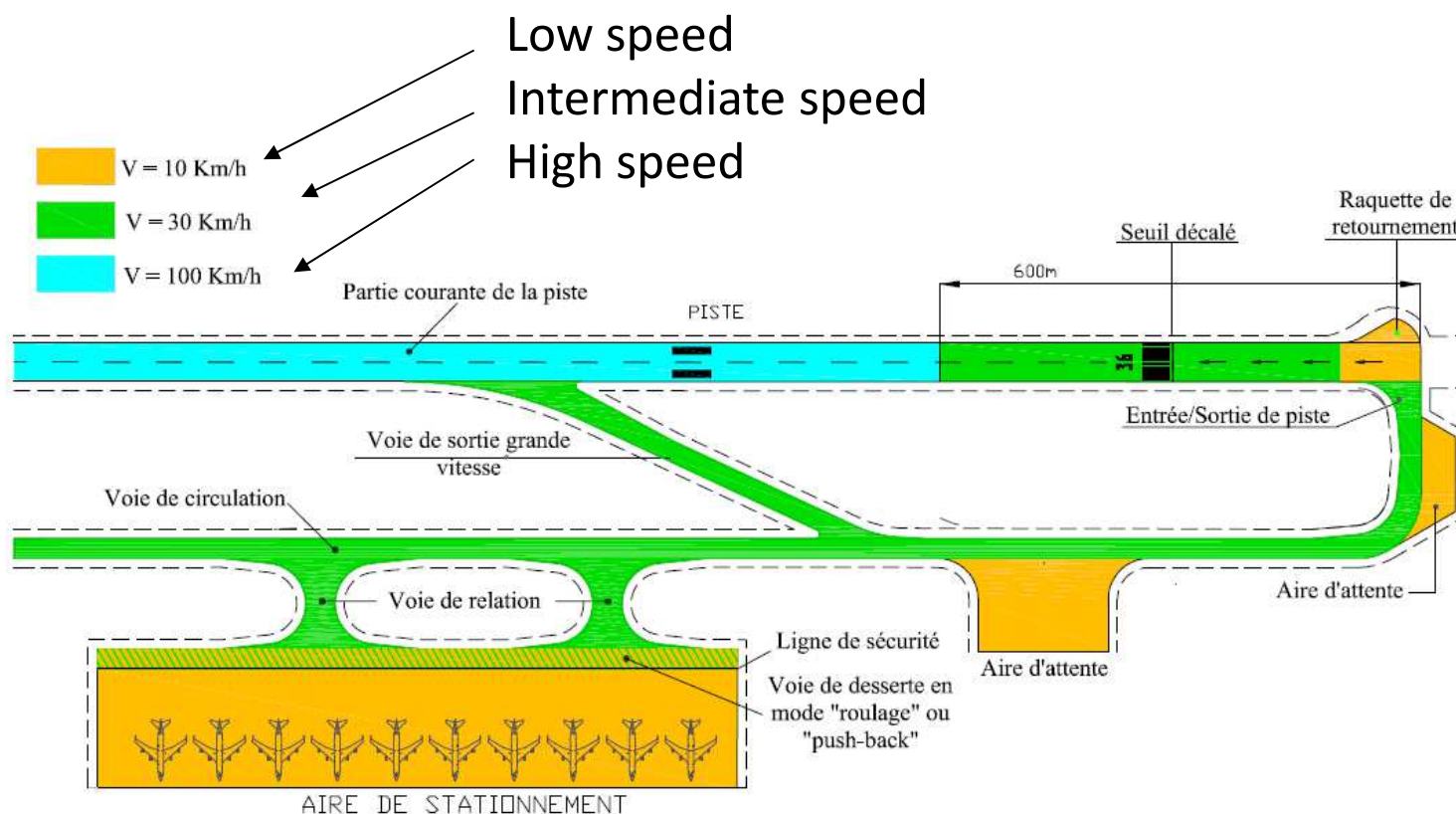
- Parameter used in the design process





Input data

Distinction of 3 types of section for the design:





Input data

- Speed

<i>Section considered</i>	<i>Speed for calculations</i>
High speed section	100 km/h
Intermediate speed section	30 km/h
Low speed section	10 km/h*

* Fictitious value used for calculation

- Speed is related to the frequency of loading of asphalt materials (the higher the speed of loading, the higher the asphalt concrete modulus value)
- It is considered that:
100 km/h ↔ 10 Hz and
$$f(Hz) = \frac{V(km/h)}{10}$$



Input data



- Lateral wander

<i>Section considered</i>	<i>Lateral wander (standard deviation)</i>
High speed section	0.75 m
Intermediate speed section	0.5 m
Low speed section	0 m

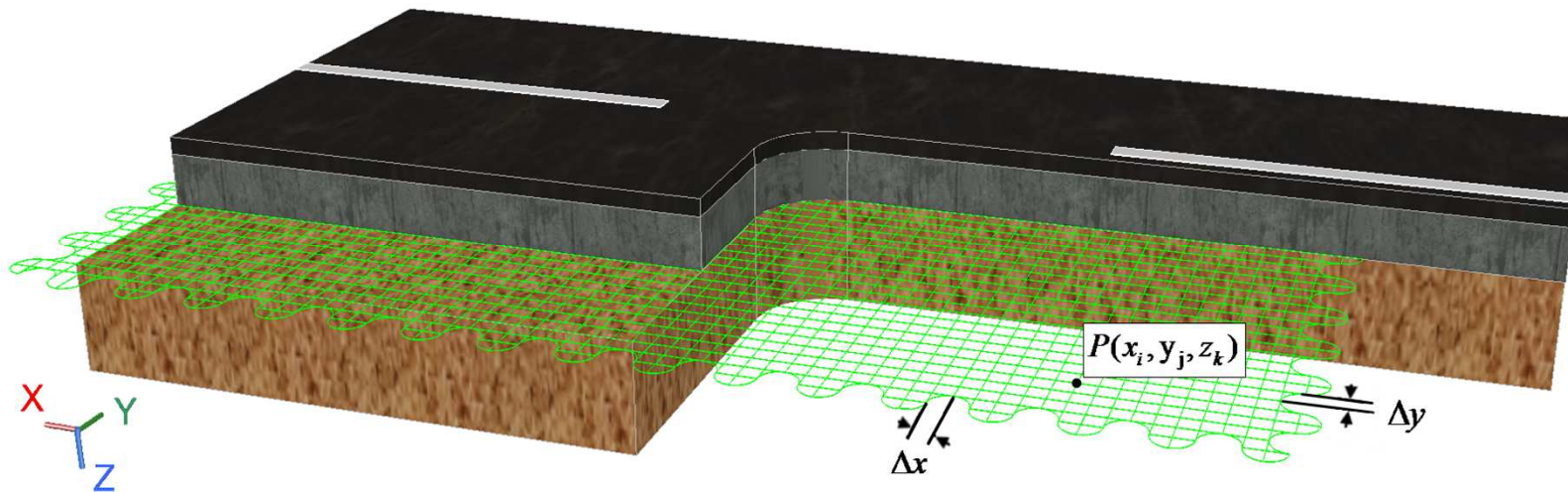
- Lateral offsets from the runway centreline are taken into account
- Offsets distribution is supposed to be Gaussian, centered and defined by its standard deviation S_{bal}



Strains and stresses calculation



- Multi-layered semi-infinite elastic model (Burmister)
- Elastic modulus (E) and Poisson's ratio (ν) assigned to each layer



- Calculation at every node of the grid of the strains/stresses under each aircraft loading

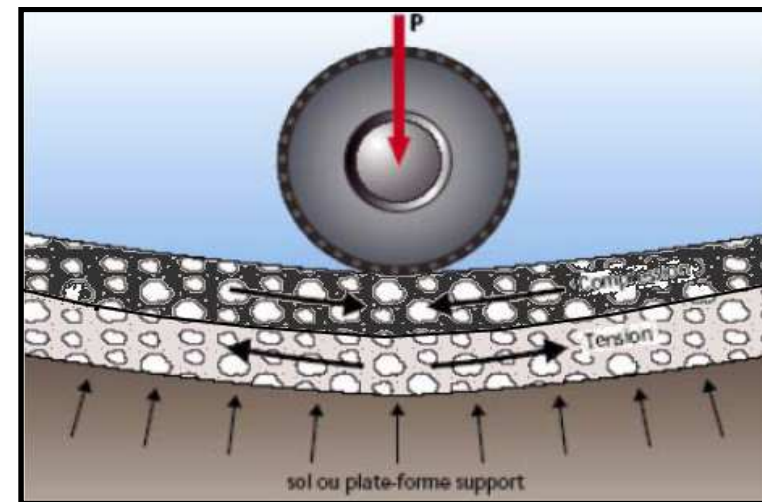


Strains and stresses calculation



- Calculations at two levels:

- Horizontal extensions ε_t at the base of the asphalt materials (fatigue)
- Vertical contractions ε_{zz} at the top of the unbound layer (permanent deformations)

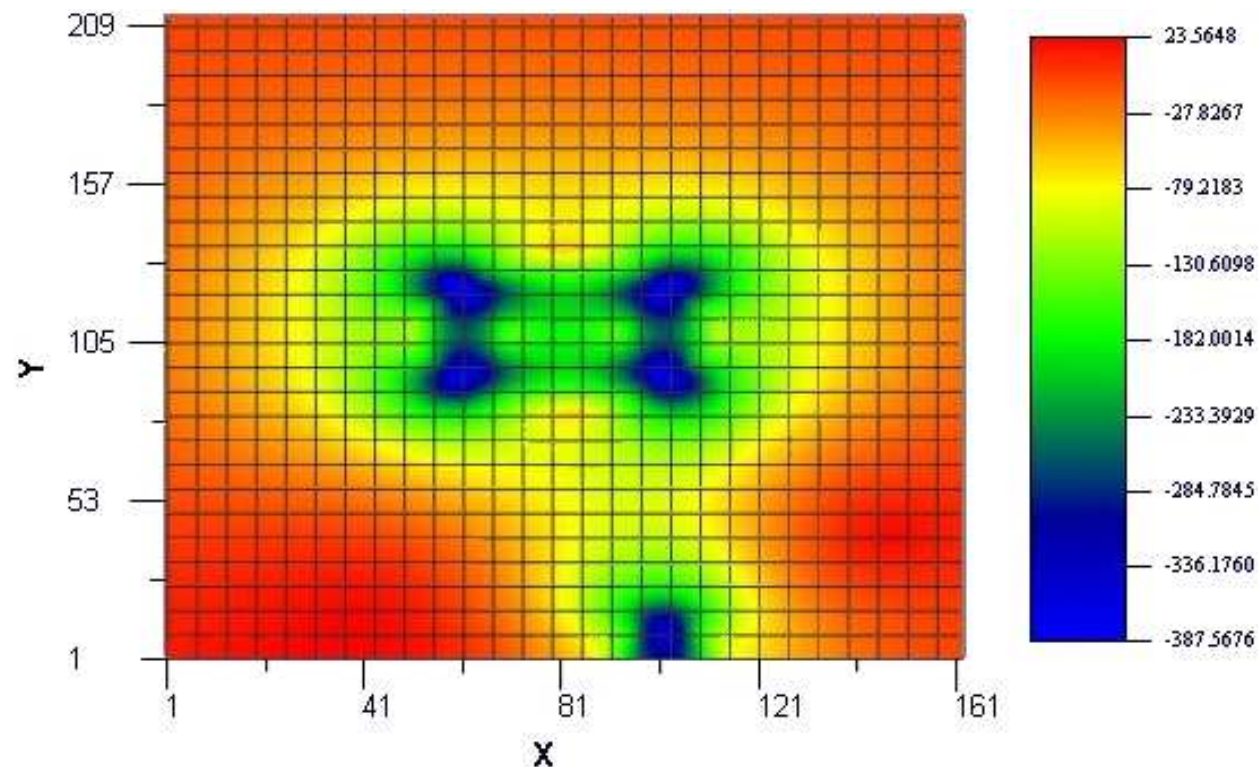




Strains and stresses calculation



- Example of fatigue tensile strain calculation for an A330-300 aircraft





Failure model



- Wöhler failure model is used
$$N = \left(\frac{K}{\epsilon_{\max}} \right)^{\beta}$$
 - N the number of passes to failure
 - $\epsilon_{\max}$ the maximum strain under an aircraft loading
 - $\beta > 1$, $b = -1/\beta$ slope of the fatigue line
 - K a constant of the material
- For granular material permanent deformations,
$$\epsilon_{\max} = \epsilon_{zz} \text{ and } \begin{cases} K = 16000 \\ \beta = 4.5 \end{cases}$$



Failure model

- For asphalt concrete fatigue, $\varepsilon_{\max} = \varepsilon_t$ and

$$K = 10^{6/\beta} k_{\theta f} k_s k_r k_c \bar{\varepsilon}_6$$

Transposition of the fatigue test performed at 10°C and 25Hz to the actual design temperature and frequency

Local bearing capacity heterogeneities of a granular layer

Scattering of the fatigue test results and of the layer thickness variation

Shift factor

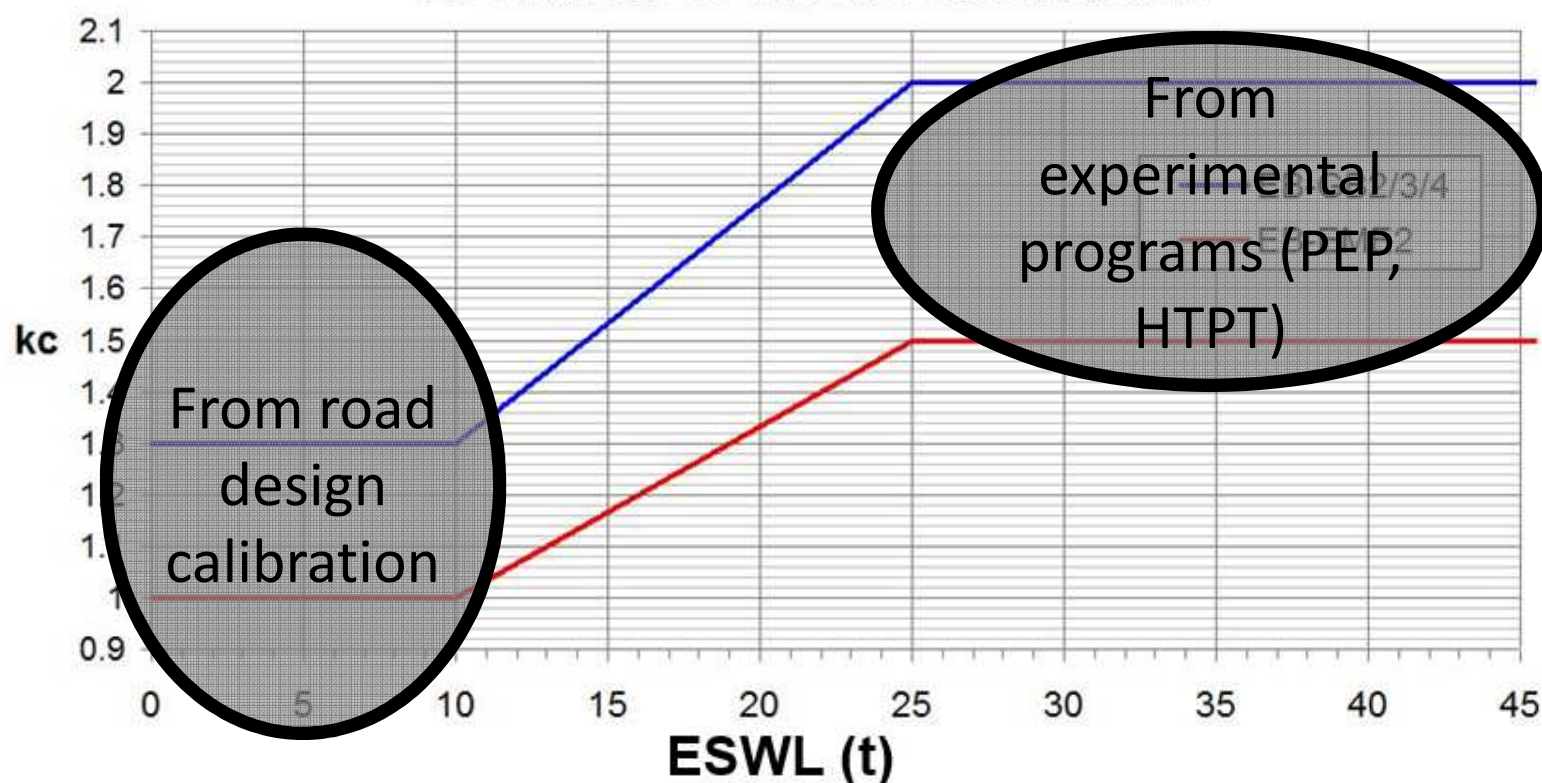
Fatigue resistance of the material



Failure model

- Shift factor k_c : difference between lab and the field

Evolution of the k_c coefficient





Damage calculation

- Elementary damage: $\Delta D = \frac{1}{N}$

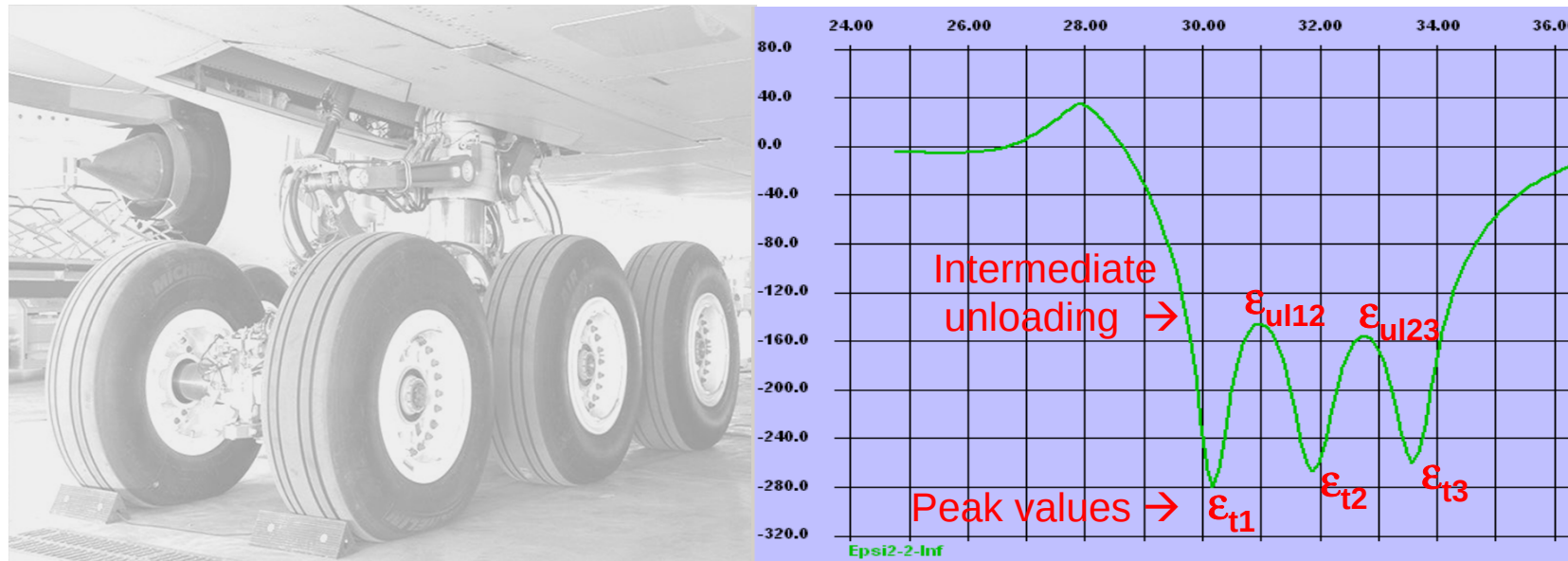
- Additivity of damage, Miner's rule:

$$D = \sum_i n_i \Delta D_i = \sum_i \frac{n_i}{N(\epsilon_{\max i})}$$

n_i : number of passes for aircraft i



Damage calculation

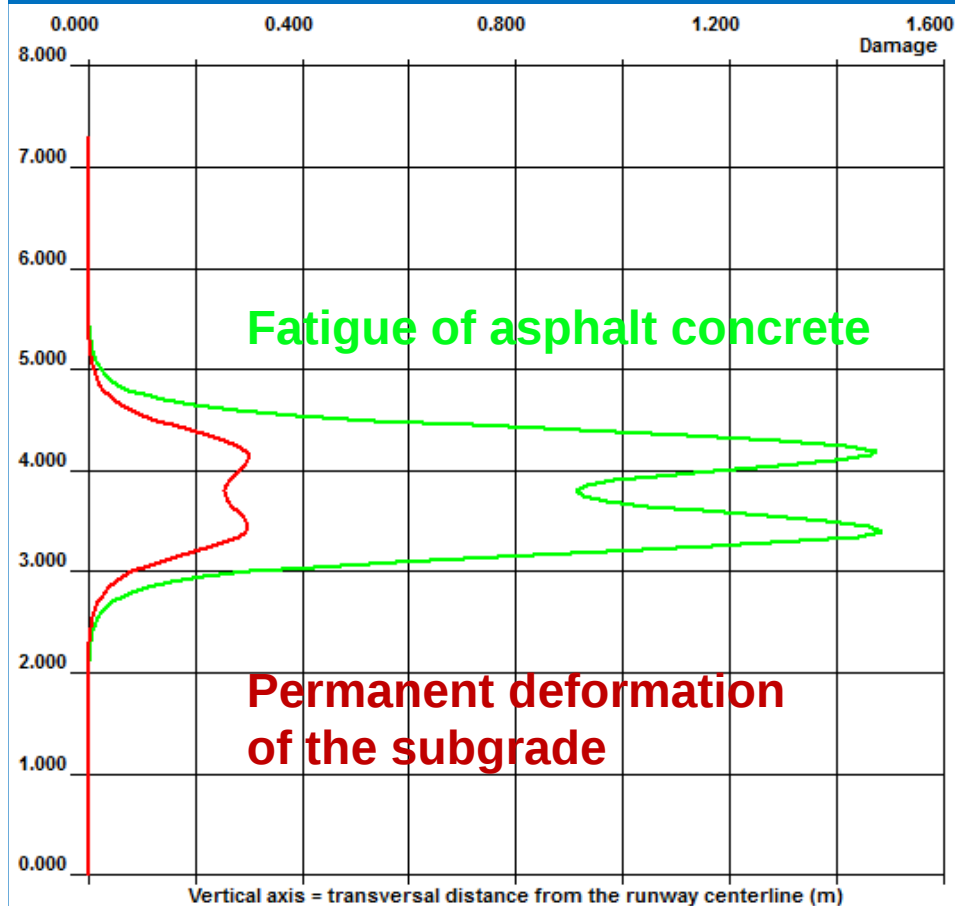


- Calculation using the time history strain profile

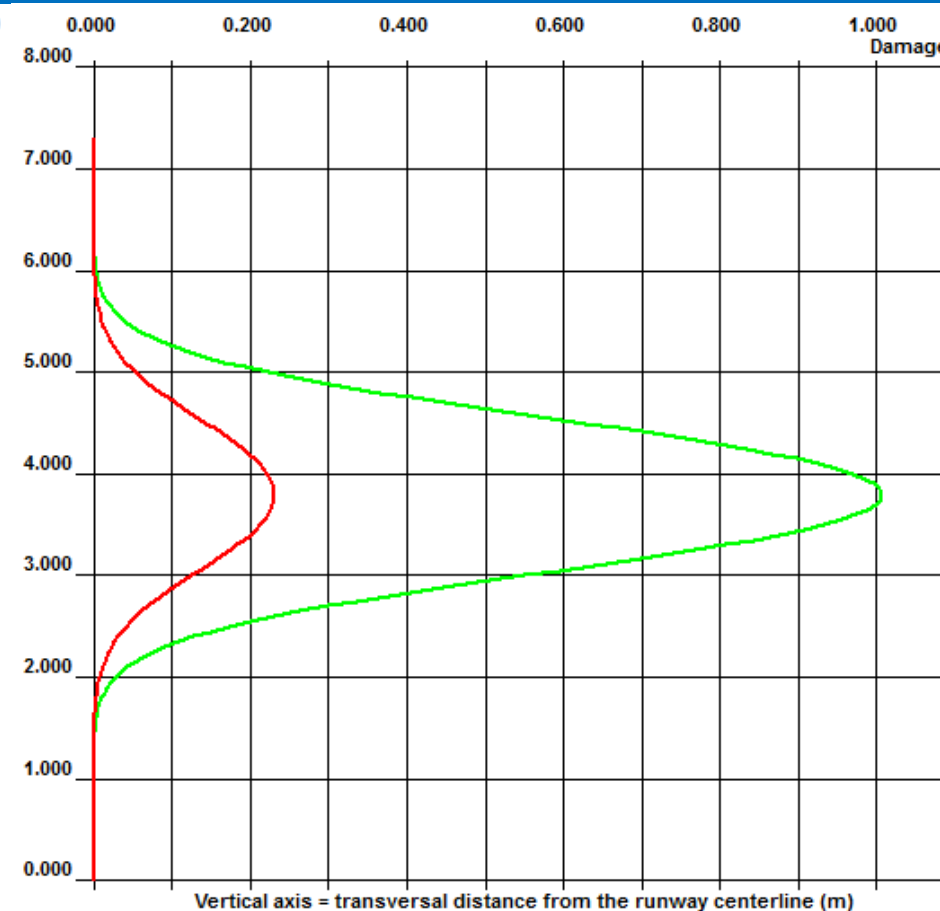
$$D_{tridem} = \frac{1}{K^{\beta}} \left(\epsilon_{t1}^{\beta} - \epsilon_{ul12}^{\beta} + \epsilon_{t2}^{\beta} - \epsilon_{ul23}^{\beta} + \epsilon_{t3}^{\beta} \right)$$



Damage calculation



Without wander



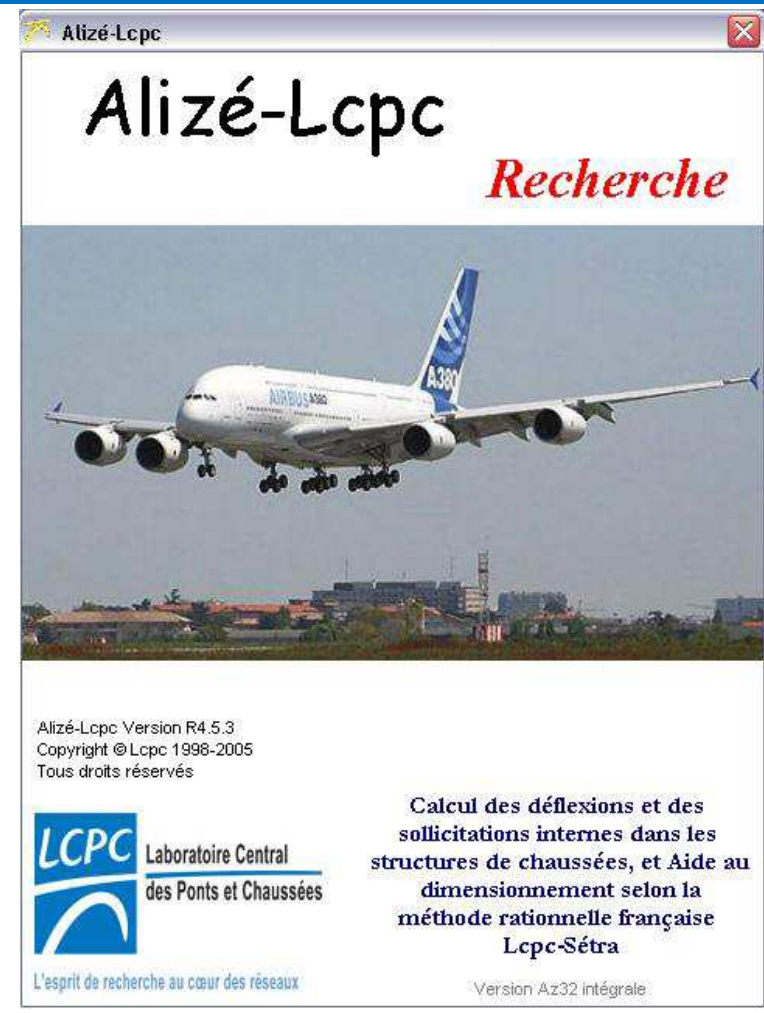
With wander



Alizé-Airfield Pavement Software



- **Alizé-Airfield Pavement:**
Developed as a module of Alizé-Lcpc software
- Available soon (2014)
- Distributed by the company ITECH





Validation of the method



- Validation by a committee of experts:
 - Main French public works companies
 - Airport managers
 - Aircraft manufacturers
- Validation based on more than 50 case studies (with various contexts)



Next steps

Pavement overlay

- Flexible on flexible (underway)
- Other cases (flexible on rigid...)

- New rigid pavement design (project starting)

- ACN/PCN methodology to be revised



Any questions ??

Thank you for your attention

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